Testing>Final inspection checklist

This is the checklist that we use in the factory: it covers every part of the aircraft and it **must** be completed before your first flight.

Time to complete properly: at least one full day, probably more. Do **not** rush these checks!

Aiı	rcraft Registration:	
Jal	piru Model:	
Pr Pr	rewall forward components opeller: Jabiru / Sensenich opeller Size: opeller Serial No:	
Pr	opeller Bolts	
1	Check torque (Jabiru prop 7 ft/lbs, Senenich prop 12ft/lbs)	
2	Check tip tracking is accurate to < 2mm, adjust if required	
3	TorqueSeal the propeller nuts	
4	Visually check propeller for serviceability	
Sp	pinner	
1	Spinner screws secure	
2	Check that the spinner and spinner backing plate has been indexed correctly and marked on the spinner and spinner flange	
3	Visually check spinner tracking with the engine running	
Er	ngine	
1	Engine Serial Number:	
2	Engine mount bolts secure:	
	a) firewall	
	b) engine	
3	Thrust washers in place and correct	
4	Fuel hose secure:	
	a) bulkhead to fuel pump	
	b) fuel pump to carburettor	
	c) fuel lines all clamped and tied	
	d) fire sleeve covering clamps and lock wired in place	
5	Breather hose from fuel pump: drain line routed correctly	
6	Drip tray under fuel pump: drain line routed correctly	

Spark Plugs/Coil leads 1 Spark plugs are secure in cylinder heads 2 Spark plug leads are secure on spark plugs 3 Spark plug leads not rubbing (on air ducts, cylinder heads) 4 Terminal nuts are secure 5 CHT sender (12mm terminal) secure 6 Spark plug leads secure in distributor caps 7 16 gauge earth wire secure and clipped to each coil **Ram Air Ducts** 1 Baffles in place 2 Cap screws secure through ducts to cylinder heads 3 Springs in place inboard side of ducts 4 Spring lock-wire correct and secure 5 Check alignment of rubber front lip with the top cowl 6 Air ducts not touching the underside of the top cowl or oil door hinge 7 Cooling tubes directed onto coils **Engine Oil** 1 Oil level correct (capacity: 2200 = 2.3 litres, 3300 = 3.5 litres) 2 Sump plug tightened and lock wired 3 Oil filter secure (spin on filter, hand pressure only) 4 Oil cooler bracket secured to engine 5 Oil cooler secured to bracket 6 Oil cooler hose clamps secure x 4 7 Check oil hose routing 8 Fire sleeve secure and lock wired over hose clamps Oil breather hose and oil bottle secure Throttle cable 1 Cable lock nuts secure on both ends of cable 2 Cable end free in carburettor link 3 Split pins secure 4 2 x 5/16 washers between cable and carburettor link 5 Cable alignment correct with outer at both ends

6 Cable adjustment

	a) full throttle b) idle	
Ch	oke Cable	
1	Lock nuts secure both ends of cable	
2	Cable end free in carburettor link	
3	Split pin secure	
4	Cable alignment correct with outer	
5	Operational	
Ca	rburettor Heat Cable	
1	Lock nuts secure both ends	
2	Cable end free in carburettor link	
3	Split pin secure	
4	Cable alignment correct with outer	
5	Operational	
Ca	bin Heat Cable	
1	Lock nuts secure both ends	
2	Butterfly sealing in ON and OFF positions	
3	Over-centre spring fitted properly and working	
4	SCAT hoses routing – nothing rubbing, etc	
Sc	at Hoses	
1	Hot air mixer box to carburettor	
	a) hose clamps secure	
	b) hose not bunched up under carburettor	
	c) cobra head fitted and aligned properly	
	d) smooth routing	
2	NACA duct in cowl to hot air mixer box cold air inlet	
	a) hose clamps secure	
	b) smooth routing and secure	
3	Heat shroud on muffler to hot air mixer box hot air inlet	
	a) hose clamps secure	
	b) smooth routing	

Nose leg and housing

1	Front tyre pressure checked: 34 psi	П
2	[Tyre Pressure] decal in place on spat above grommet	_
3	Nose wheel - ease of rotation	
4	Nose leg – lubricated and ease of rotation left to right	
5	Assembly – all bolts secure in nose leg	
6	Housing bolts secure	
7	Rod ends secure both sides	
8	Safety hole in rod ends within limits	
9	Jam nuts on rod ends secure	
10	Air valve grommet in place	
11	Spat retaining screws secure	
12	Centring springs do not become coil bound full deflection	
Ele	ectrical	
1	Battery tie down secure	
2	Battery terminals secure	
3	Battery earth secure on firewall	
4	Starter solenoid locknuts and wires secure	
5	Regulator wires secure and correct	
6	Temp sender wires secure (EOT and CHT) plus EGT if fitted	
7	All wiring cable tied and secure	
8	Oil pressure wire correct and secure (G terminal)	
Со	owls	
1	Top and bottom cowl alignment correct (no gaps)	
2	Retained nuts secure and correct	
3	Oil door operational	
4	Spinner not touching front of cowl and Propeller	
5	Oil cooler not touching bottom cowling	
6	Cylinder heads not touching top or bottom cowling	
7	NACA duct secure with no sharp edges	
8	Cam locks secure and operational	

Interior

Fuel system

1	Fu	el filter secure and flow direction correct		
2	Fu	el pump (electronic) secure and flow directi	on correct	
3	Fu	el lines secure:		
	a)	fuel pickups and breather lines from wing	roots to header tank inlets	
	b)	header tank outlet to fuel filter		
	c)	fuel filter to electric fuel pump		
	d)	electric fuel pump to fuel tap on console		
	e)	fuel tap on console to bulkhead fitting		
4	Fu	el line sheathing in place and secure		
5	Ch	eck fuel valve mounting and lever secure to	console	
6	Fu	el Breather:		
	a)	check breather hose clamps and breather	route	
	b)	wing breather tube to wing root secure		
	c)	check ball in breather in wing tank cap is fr	ee	
	d)	check fuel breather tube is secure to body	fuel filler	
	e)	fuel breather check has been conducted:		
		Note: you must be able to hear air in all ou	tlets independently by:	
		i)Removing filler cap in left wing and blow	into tank	
		ii) Removing filler cap in right wing and b	low into tank	
Fu	el g	gauges	LEFT	RIGHT
1	Wi	ndow fitted in fuselage opposite wing roots		
2	Ca	libration decal in place left and right		
3		res connected to left and right senders		
4		t sender drives left gauge and vice versa		
5		leaks from sender boss	Ц	Ш
		low test		_
1		litres of fuel in each wing tank	ON : OFF :ff	
2		eck fuel tap has been correctly indexed i.e. ect fuel tap on and measure the fuel flow (r		⊔ Vaec tack)
J		electric fuel pump OFF lit		
	,	electric fuel pump ON lit		
	Th	ere should be around 1 litre per minute flow		ON

Controls

Rudder pedals

1	Rudder springs installed properly	
2	Rudder stops lining up properly	
3	2 x Pedal mount bolts secure (pushrod to pedals)	
4	Rudder pedal keepers secure and correct	
5	$2 \times \%$ " and $4 \times 3/16$ " bolts secure in bearing blocks	
6	2 x ¼" bolts secure in pushrods (pushrod to nose leg yoke)	
7	Check jam nuts on rod ends secure x 5	
8	Check safety hole is within limits on rod ends x 5	
9	Check rod end free play at full deflection, both directions	
10	Rudder clear of fin channel. Rudder cannot bind on channel	
Ru	dder cable	
1	Anchor secure and correct position in clamp	
2	Rod end secure in pedal with ¼" washer outside	
3	Safety hole is within limits on rod ends	
4	Jam nuts on rod ends secured	
5	Rod end free play at full deflections, in both directions	
6	Rudder pedal stops adjusted and secure: rudder should move 78mm left and right from the 5mm offset centre mark	
Tri	im	
1	Trim levers (left and right) are secure	
2	Lever pivot secure and split pin fitted through nut	
3	Free play between left and right side levers not excessive	
4	Friction correct and secure	
5	Rod end secure	
6	Safety hole is within limits on rod ends	
7	Jam nuts on rod ends secured	
8	Cable position in clamp correct and anchor secure	
9	Operation full forward and full back trim	
10	Spring not coil bound with elevator fully deflected up	
11	Placard fitted to console top in correct sense	

Brake 1 Handle assembly correct and secure Check safety hole within limits 3 Operation 4 Free play 5 Fluid level full 6 Stainless striker in place 7 Handbrake correct and operational **Throttle** 1 Cable anchor secure 2 Cable alignment correct 3 Cable end free in output lever 4 Output lever secure 5 Input lever x 2 secure 6 Throttle lock hole in pilot's input lever 7 Bearing blocks x 2 secure 8 Throttle stops correct and secure Main control shaft 1 Control stick pivot secure 2 Yoke secure (no free play) 3 Rod end secure 4 Safety hole is within limits on rod ends 5 Jam nuts on rod ends secured 6 Elevator cable anchor secure and correct position in clamp Elevator cable clear of centre console in all positions

(particularly elevator UP and aileron to left hand side)

9 Aileron bell crank lines up properly with stops and is not binding

8 Rod end free at full deflections

Ai	leron Cable LEFT RIGHT		
1	Cable anchor wedges in place		
2	"R" clips in place		
3	Rivets x 3 TLR through each bracket and seat back		
4	Rod end secure		
5	Safety hole is within limits on rod ends		
6	Jam nuts on rod ends secured		
7	Aileron stoppers secure x 2 (cannot over centre aileron bellcrank)		
Se	eat belts LEFT RIGHT		
1			
1	Check bolts are through Nyloc and buckle has free play		
2	Check bolts are through Nyloc and buckle has free play Operation of buckle and length adjustments play		
-		_	
2	Operation of buckle and length adjustments play		
2	Operation of buckle and length adjustments play Belts not twisted play		
2 3 4	Operation of buckle and length adjustments play Belts not twisted play Seat belt anchor secure (lower and upper) play		
2 3 4 5	Operation of buckle and length adjustments play Belts not twisted play Seat belt anchor secure (lower and upper) play		

Instrument panel 1 Secure and level 2 Instruments square (check that the slip/skid ball is centred). 3 Slip/skid ball air bubble not visible 4 Circuit breakers secure and correctly labelled 5 Switches secure and correctly labelled 6 Registration number displayed on panel 7 Mag switch guard in place and square with rounded edges 8 All instruments fitted and operational * 9 Instrument markings correct 10 Altimeter QNH correct 11 All avionics installed and operational П 12 Cigarette lighter / 12v power fitting installed * Fit 1 metre of rolled up surgical tubing to the brass static tube connector under the panel. Watch the pressure instruments while slowly unrolling the tubing. The altimeter should start to show an increase in altitude. The vertical speed indicator should start to show a positive rate of climb and the airspeed indicator may show a small positive indication. When the altimeter shows 2000 feet pinch off the tubing. There will be some initial decrease in altitude and the vertical speed indicator will read zero. The altimeter should then hold the indicated altitude for at least one minute. If altitude is lost then look for leaks. Note: NEVER apply suction orally or mechanically to a static line - it can damage the instruments. **Flaps** 1 Mounting bolts secure 2 Rod end bolt secure: motor end and flap shaft ends 3 Rod end not binding in any position 4 Safety hole is within limits on rod end 5 Jam nut on rod end secured 6 Visual on input switch (knob shape, operation and decal) 7 Visual on bearing blocks 8 Check all nuts are in safety Flap indicator 1 Secure and correct indication: a) Flaps up b) First stage: half flap

c) Second stage: full flap

U	pholstery:		
1	Fitted and clean		
2	Seat covers		
3	Map pockets		
4	Flap shaft upholstered		
5	Centre console upholstered		
6	Fuel line covers in place		
Pl	acards and decals:		
1	[No Smoking] on panel		
2	[No Intentional Spins] on panel		
3	[Voltage must be displayed at all times] above EMS (if applica	ble)	
4	[Transponder not calibrated] (if applicable)		
5	Switches labelled (label above switch, [OFF] below switch)		
6	[Aircraft built to LSA standards] decal beside main placard		
7	Warning placard above windscreen. Weight entered		
8	Baggage (Limit, loading rules, seat backs)		
9	Loading stations		
Do	oors: LEFT RIGHT		
1	Operation: open and shut, latches working		
2	Restraining straps secure and correct length		
3	Hinge and pin secure		
4	Catches secure and freely operating (main plus top latch)		
5	Visual on door jamb		
6	Visual on door seal		
7	Visual on handle		
8	Visual on top catch		
9	Decals		
	a) [Do Not Lean On Door]		
	b) [Exit]		
	c) [Open]		

Exterior

W	ings: LEFT RIGHT	
1	Root bolts inspected (correct tension) x 2	
2	Strut bolts x 2 inspected, correct tension	
3	Warning decals in wing roots x 2 [Do Not Tighten]	
4	Wing root cover strips secure	
5	Upper strut fairing secure and clear of strut	
6	Lower wing strut/belly fairings secure	
Fla	aps	
1	Hinge points secure with pivot bolts in safety	
2	Pushrod rod ends secure and not binding in any position	
3	Safety hole in rod ends is within limits	
4	Jam nuts on rod ends secure	
5	Pushrods not fouling on cover strips	
6	Drive shaft output secure	
7	Drain holes have been drilled	
8	Flap operation:	
	a) flaps up	
	b) first stage – half flap	
	c) second stage – full flap, not fouling fuselage	
9	Flap slot gap even, smooth and the same size on both wings	
Fu	el caps	
1	Filler caps secure in fuel filler body	
2	Earth wires secure from caps to breather fitting in tank	
3	Quick drain: operation and lock wired	
4	Decals in place and correct	

1 Deflections a) Neutral (level with lower wing skin) with stick centred degrees, or touches template c) DOWN _____degrees 2 Hinge pins and pin retainers secure 3 Rod ends secure 4 Safety hole on rod ends within limits 5 Jam nuts on rod ends secure 6 Cable anchor secure and correct position in clamp 7 Drain holes a) Wing drain hole in trailing edge b) Aileron cable inspection windows secure **Pitot** 1 Secure 2 ASI operation (see * below) 3 Cover present 4 Pressure Test with no leaks (see * below) * Slip 1 metre of surgical tubing over the pitot tube. Have one person watch the airspeed indicator while the other person very slowly rolls up the tubing. This will apply pressure to the instrument in a gentle and controllable manner. When the ASI needle reaches cruising speed pinch the tube shut – the ASI should hold that reading for at least one minute. A fast drop off will indicate a leak in the instrument, fitting, lines or the test hose attachment. Note: NEVER blow into a pitot tube – it can damage the ASI. **Stall warning:** 1 Stall warning horn secured in cabin 2 Operation correct (suck to ensure reed is working)

Ailerons

LEFT RIGHT

3 Gauze and lip installed correctly and clear

Main Undercarriage Main gear legs secure Stub axle secure 3 Hubs secure 4 Stub axle extension is secure 5 Calliper secure 6 Discs secure 7 Ease of rotation 8 Tyre pressure 45 psi 9 [Tyre Pressure] decal in place 10 Brake line not rubbing on spat 11 Spats

LEFT RIGHT

a) screws secure

b) clear of tyre

c) retain nuts

d) outer edge – nothing rubs

e) air valve grommet in place

f) [No Step] decal in place

Empennage

Elevator Deflections

1	UP	degrees, or touches the UP template	
2	DO	WN degrees, or touches the DOWN template	
3	2m	m clearance to the stop on fuselage at full DOWN deflection	
Ele	va	tor	
1	Hir	nge pins and retainers secure	
2	Ro	d ends secure	
3	Saf	ety in rod ends within limits	
4	Jan	n nuts on rod ends secure	
5	Cal	ble anchor secure and correct position in clamp	
6	Dri	ve plate secure	
7	Ro	d end free play at full deflections	
8	End	d-plug not touching end-cap full deflection down	
9	Ele	vator min 2mm clear of stabiliser channel	
10	Ele	vator cannot bind on horizontal stabiliser channel	
Tri	m		
1	Piv	ots secure	
2	Spr	ring rod	
	a)	split pins secure	
	b)	springs lubricated	
	c)	black block secure and free	
	d)	rod ends secure	
	e)	safety hole within limit	
	f)	jam nuts secure	
	g)	rod end free at full deflections	
3	Tri	m cable	
	a)	cable anchor secure and correct position in clamp	
	b)	rod ends secure	
	c)	safety hole in rod ends within limits	
	d)	jam nuts on rod ends secure	
	e)	rod end free at full deflections	

Ve	entral Fin:	
1	Retained nuts in fuselage secure	
2	Elevator horn not hitting cut-out in ventral fin at full up deflection	
3	Inspection holes drilled	
4	Drain holes drilled out	
5	Ventral fin secure onto fuselage	
Rι	ıdder:	
1	Centre 5mm offset to right when pedals level	
2	Deflection	
	a) Left	
	_mm from the offset centre mark	
	b) Right	
	_mm from the offset centre mark	
3	2mm clearance to the stops on fuselage at full deflection	
4	Hinge pin and pin retainers secure	
5	Rod end secure	
6	Safety in rod end within limits	
7	Jam nut on rod end secure	
8	Cable anchor secure and correct position in clamp	
9	Rod end free at full deflections	
St	atic Vent:	
1	Tip secure	
2	Holes clear	
3	Decal in place	
	ote that leak testing of the static system has been performed as part of t ecklist.	he instrument
۱h	ereby certify that I have completed this Final Inspection Checklist:	
Ov	vner / Builder / Engineer (strike out any that do not apply)	 Date
- •	,, U (